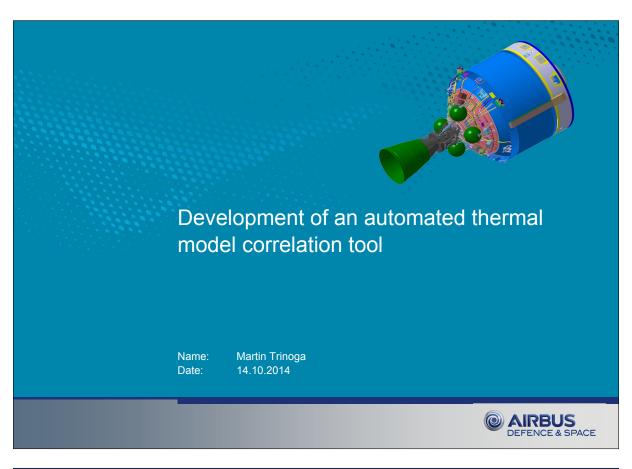
Appendix P

Development of an automated thermal model correlation tool

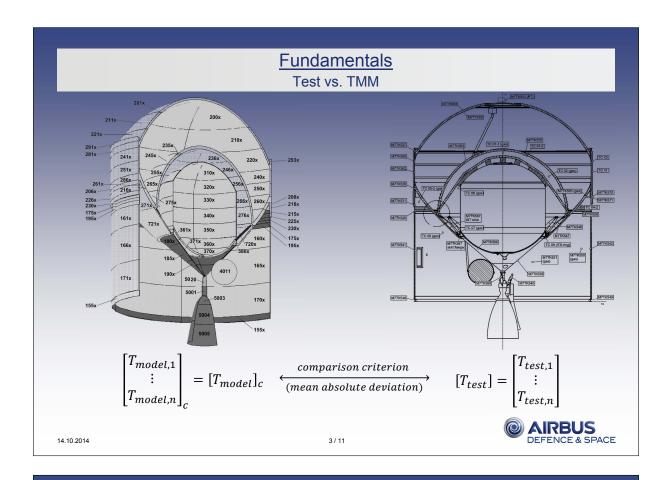
Martin Trinoga (Airbus Defence and Space, Germany)

Abstract

A generally essential part in the development of a spacecraft is the implementation of a thermal model for the temperature predictions during the operation phase. In order to achieve a good accuracy of the assessed temperatures, the thermal model has to be correlated with measurements from predetermined thermal tests. Currently this correlation is only possible in a manual way at Airbus DS GmbH. As this manual method is often a time consuming process, a new method for solving this problem is currently under development. With this newly developed MATLAB® tool (TAUMEL = Tool for AUtomated Model correlation using Equation Linearization) it will be possible to correlate restricted thermal models from ESATAN-TMS automatically. The first master thesis regarding this automated approach was finished in December 2013, whose outcome will be presented. Because of the promising potential of this new correlation method, Airbus DS in Bremen is currently working within the frame of a second master thesis. The current status will be presented as well.



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<u>Fundamentals</u>

Parametric correlation

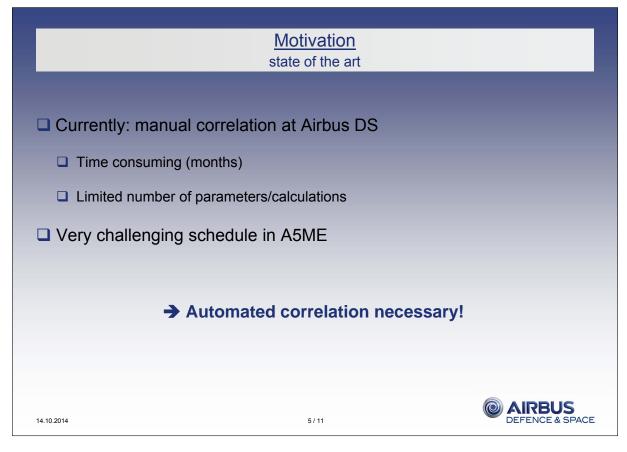
$$C_{i} \frac{dT_{i}}{dt} = \sum_{\substack{i=1\\j\neq i}}^{N} GL_{ij} PL_{ij} (T_{j} - T_{i}) + \sigma \sum_{\substack{i=1\\j\neq i}}^{N} GR_{ij} PR_{ij} (T_{j}^{4} - T_{i}^{4}) + \dot{Q}_{i}$$

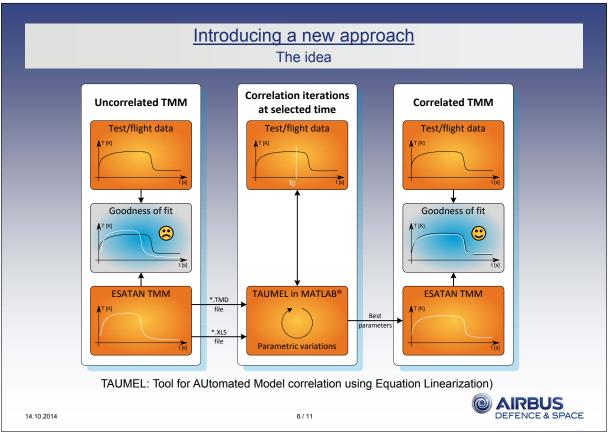
Uncertainties:

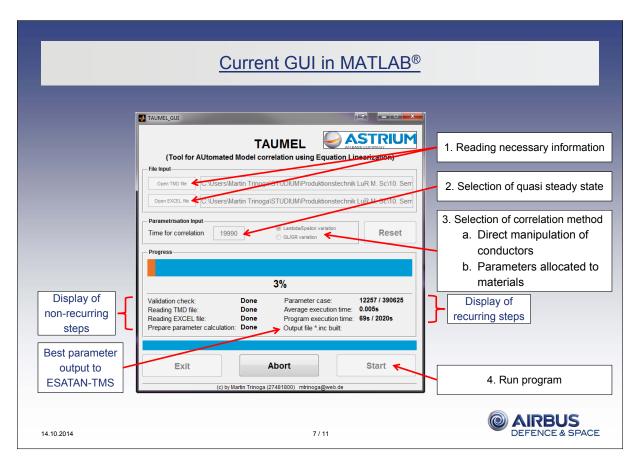
- □ Heat conduction $(GL_{ij} = \lambda_i \frac{A_{ij}}{a_{ij}})$: $\lambda: \pm 20\%$
- □ Heat radiation ($GR_{ij} = A_i \varphi_{ij} \varepsilon_i \alpha_j$): ε , α : ±20%
- $lue{}$ Convection (GF) excluded in the current development stage

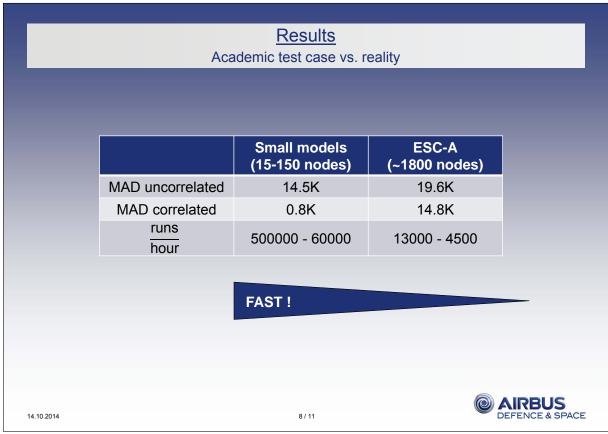
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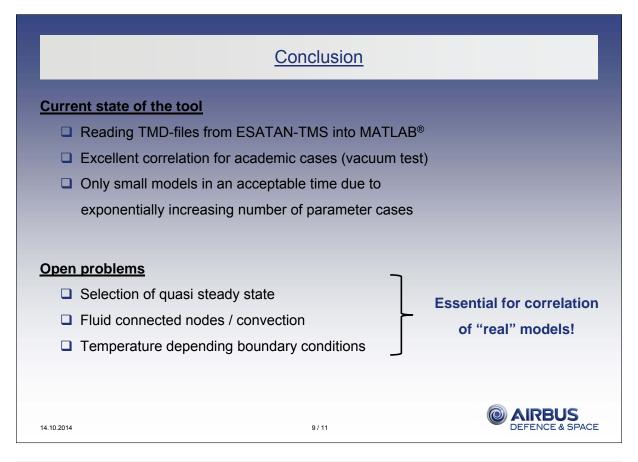


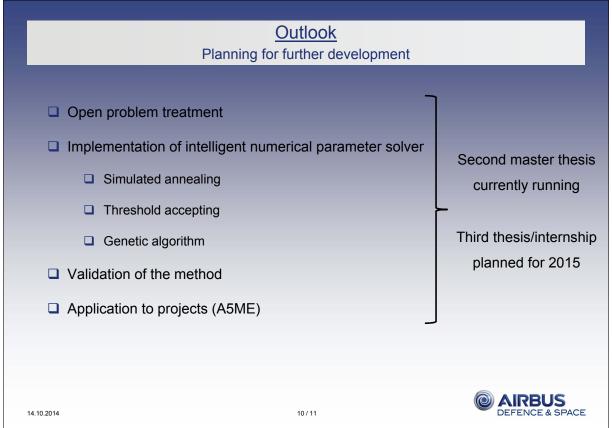




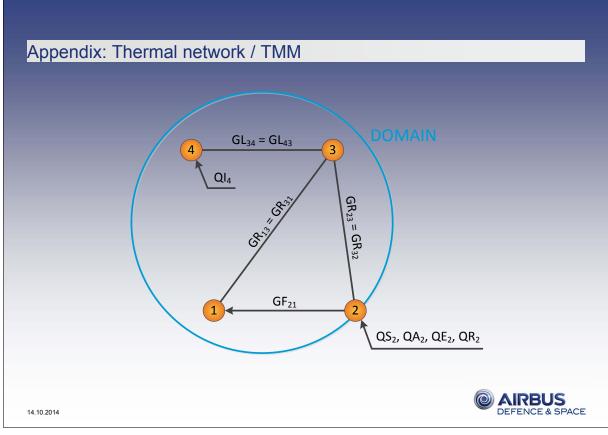












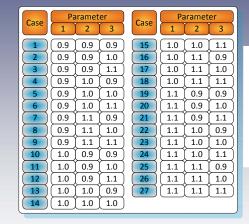
Appendix: Mean absolute deviation

$$MAD = \frac{1}{n} \sum_{i=1}^{n} |T_{test,i} - T_{model,i}|$$

14.10.2014



Appendix: Parameter combinations



14.10.2014



